

Mechanical Vibrations

Lecture 17

CHAPTER 4 **Vibration Under General Forcing** **Conditions**

Dr. Juma Yousuf Alaydi
Mechanical Engineering
Islamic University

Response to an Impulse

$$\text{Impulse} = F \Delta t = m \dot{x}_2 - m \dot{x}_1$$

$$f = \lim_{\Delta t \rightarrow 0} \int_t^{t+\Delta t} F dt = F \Delta t = 1$$

The unit impulse, $f = 1$, acting at $t = 0$,

$$m \ddot{x} + c \dot{x} + kx = 0$$

$$x(t) = e^{-\zeta \omega_n t} \left\{ x_0 \cos \omega_d t + \frac{\dot{x}_0 + \zeta \omega_n x_0}{\omega_d} \sin \omega_d t \right\}$$

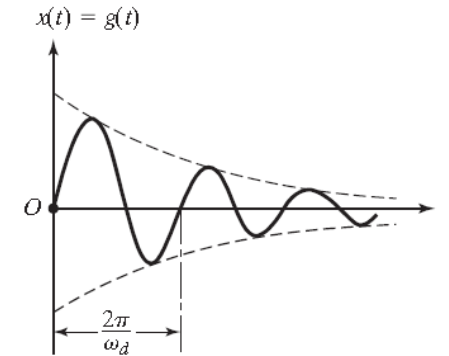
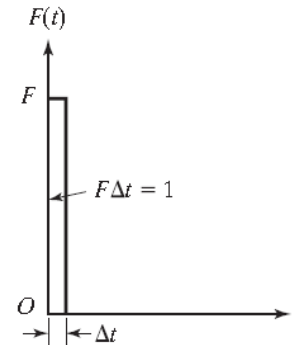
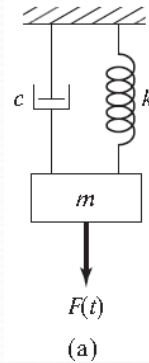
$$\text{Impulse} = f = 1 = m \dot{x}(t = 0) - m \dot{x}(t = 0^-) = m \dot{x}_0$$

$$x(t = 0) = x_0 = 0$$

$$\dot{x}(t = 0) = \dot{x}_0 = \frac{1}{m}$$



$$x(t) = g(t) = \frac{e^{-\zeta \omega_n t}}{m \omega_d} \sin \omega_d t$$



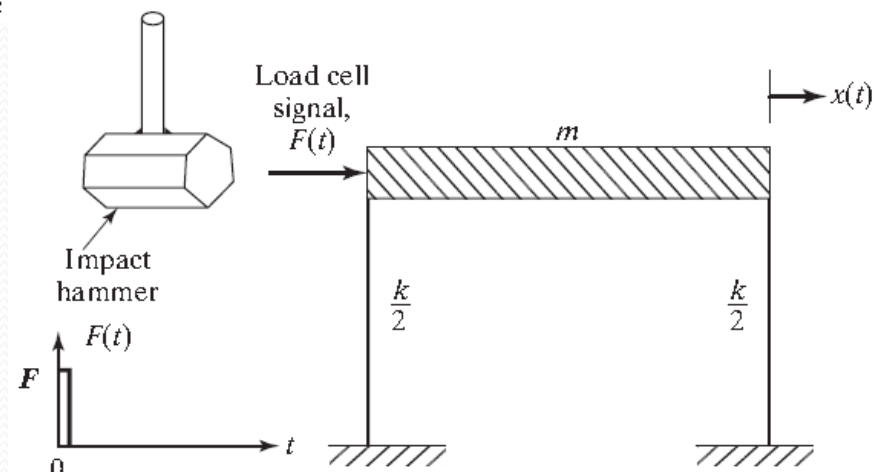
If the magnitude of the impulse is F instead of unity, the initial velocity \dot{x}_0 is F/m and the response of the system becomes

$$x(t) = \frac{F e^{-\zeta \omega_n t}}{m \omega_d} \sin \omega_d t = F g(t) \quad (4.26)$$

Response of a Structure Under Impact

EXAMPLE 4.7

In the vibration testing of a structure, an impact hammer with a load cell to measure the impact force is used to cause excitation, as shown in Fig. 4.8(a). Assuming $m = 5 \text{ kg}$, $k = 2000 \text{ N/m}$, $c = 10 \text{ N-s/m}$, and $F = 20 \text{ N-s}$, find the response of the



Solution: From the known data, we can compute

$$\omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{2000}{5}} = 20 \text{ rad/s}, \quad \zeta = \frac{c}{c_c} = \frac{c}{2\sqrt{km}} = \frac{10}{2\sqrt{2000(5)}} = 0.05$$

$$\omega_d = \sqrt{1 - \zeta^2} \omega_n = 19.975 \text{ rad/s}$$

Assuming that the impact is given at $t = 0$, we find (from Eq. (4.26)) the response of the system as

$$\begin{aligned} x_1(t) &= F \frac{e^{-\zeta\omega_n t}}{m\omega_d} \sin \omega_d t \\ &= \frac{20}{(5)(19.975)} e^{-0.05(20)t} \sin 19.975t = 0.20025e^{-t} \sin 19.975t \text{ m} \end{aligned} \quad (\text{E.1})$$

Response of a Structure Under Double Impact

EXAMPLE 4.8

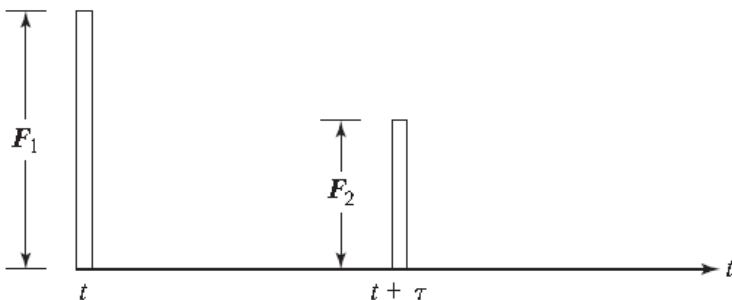
In many cases, providing only one impact to the structure using an impact hammer is difficult. Sometimes a second impact takes place after the first, as shown in Fig. 4.8(b), and the applied force, $F(t)$, can be expressed as

$$F(t) = F_1 \delta(t) + F_2 \delta(t - \tau)$$

where $\delta(t)$ is the Dirac delta function and τ indicates the time between the two impacts of magnitudes F_1 and F_2 . For a structure with $m = 5$ kg, $k = 2000$ N/m, $c = 10$ N-s/m and $F(t) = 20 \delta(t) + 10 \delta(t - 0.2)$ N, find the response of the structure.

Solution: From the known data, we find $\omega_n = 20$ rad/s (see the solution for Example 4.7), $\zeta = 0.05$, and $\omega_d = 19.975$ rad/s. The response due to the impulse $F_1 \delta(t)$ is given by Eq. (E.1) of Example 4.7, while the response due to the impulse $F_2 \delta(t - 0.2)$ can be determined from Eqs. (4.27) and (4.26) as

$$x_2(t) = F_2 \frac{e^{-\zeta \omega_n (t-\tau)}}{m \omega_d} \sin \omega_d (t - \tau) \quad (\text{E.1})$$



Eq. (E.1) becomes

$$\begin{aligned} x_2(t) &= \frac{10}{(5)(19.975)} e^{-0.05(20)(t-0.2)} \sin 19.975(t - 0.2) \\ &= 0.100125 e^{-(t-0.2)} \sin 19.975(t - 0.2); \quad t > 0.2 \end{aligned} \quad (\text{E.2})$$

Using the superposition of the two responses $x_1(t)$ and $x_2(t)$, the response due to two impacts, in meters, can be expressed as

$$x(t) = \left\{ \begin{array}{l} 0.20025 e^{-t} \sin 19.975 t; \quad 0 \leq t \leq 0.2 \\ 0.20025 e^{-t} \sin 19.975 t + 0.100125 e^{-(t-0.2)} \sin 19.975(t - 0.2); \quad t > 0.2 \end{array} \right\} \quad (\text{E.3})$$

Response to a General Forcing Condition

since the impulse is $t - \tau$, so the response of the system at t due to this impulse alone is given by Eq. (4.27) with $F = F(\tau) \Delta\tau$:

$$\Delta x(t) = F(\tau) \Delta\tau g(t - \tau)$$

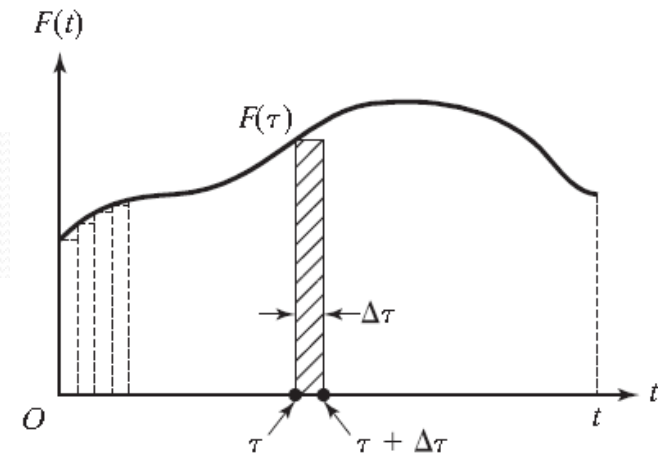
$$x(t) \simeq \sum F(\tau) g(t - \tau) \Delta\tau$$

Letting $\Delta\tau \rightarrow 0$ and replacing the summation by integration, we obtain

$$x(t) = \int_0^t F(\tau) g(t - \tau) d\tau$$

By substituting Eq. (4.25) into Eq. (4.30), we obtain

$$x(t) = \frac{1}{m\omega_d} \int_0^t F(\tau) e^{-\zeta\omega_n(t-\tau)} \sin \omega_d(t - \tau) d\tau$$



EXAMPLE 4.9

Step Force on a Compacting Machine

A compacting machine, modeled as a single-degree-of-freedom system, is shown in Fig. 4.10(a). The force acting on the mass m (m includes the masses of the piston, the platform, and the material being compacted) due to a sudden application of the pressure can be idealized as a step force, as shown in Fig. 4.10(b). Determine the response of the system.

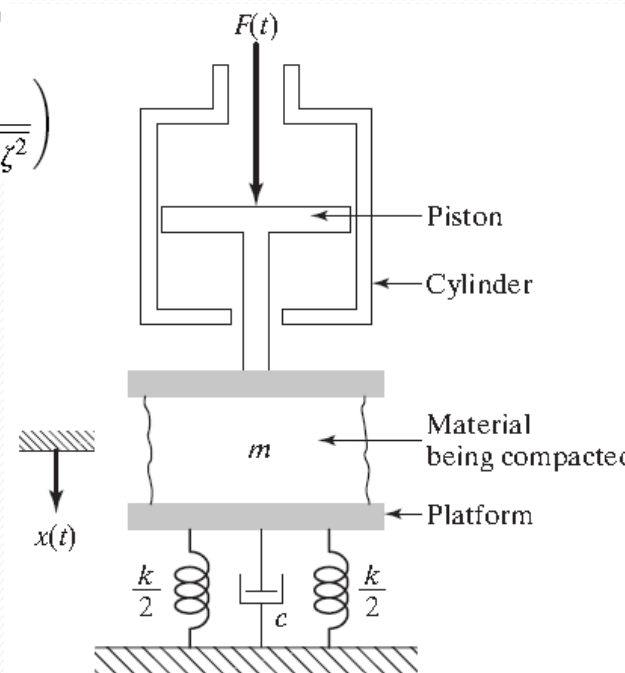
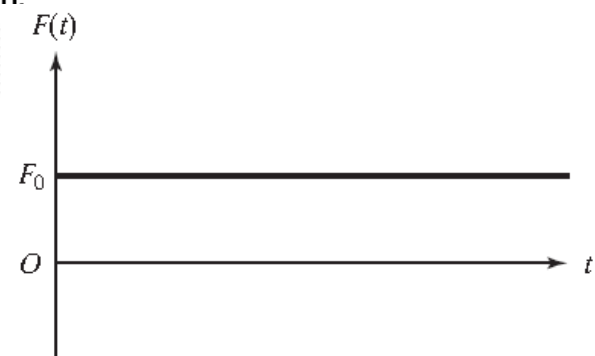
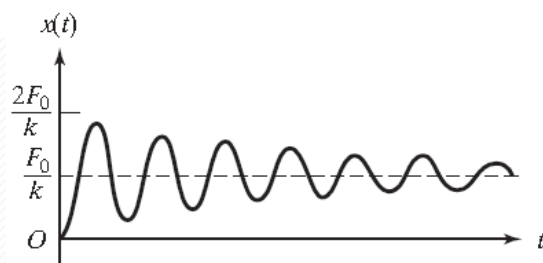
noting that $F(t) = F_0$, we can write Eq. (4.31) as

$$\begin{aligned} x(t) &= \frac{F_0}{m\omega_d} \int_0^t e^{-\zeta\omega_n(t-\tau)} \sin \omega_d(t-\tau) d\tau \\ &= \frac{F_0}{m\omega_d} \left[e^{-\zeta\omega_n(t-\tau)} \left\{ \frac{\zeta\omega_n \sin \omega_d(t-\tau) + \omega_d \cos \omega_d(t-\tau)}{(\zeta\omega_n)^2 + (\omega_d)^2} \right\} \right]_{\tau=0}^t \\ &= \frac{F_0}{k} \left[1 - \frac{1}{\sqrt{1-\zeta^2}} \cdot e^{-\zeta\omega_n t} \cos(\omega_d t - \phi) \right] \quad \phi = \tan^{-1} \left(\frac{\zeta}{\sqrt{1-\zeta^2}} \right) \end{aligned}$$

This response is shown in Fig. 4.10(c). If the system is undamped ($\zeta = 0$ and $\omega_d = \omega_n$), Eq. (E.1) reduces to

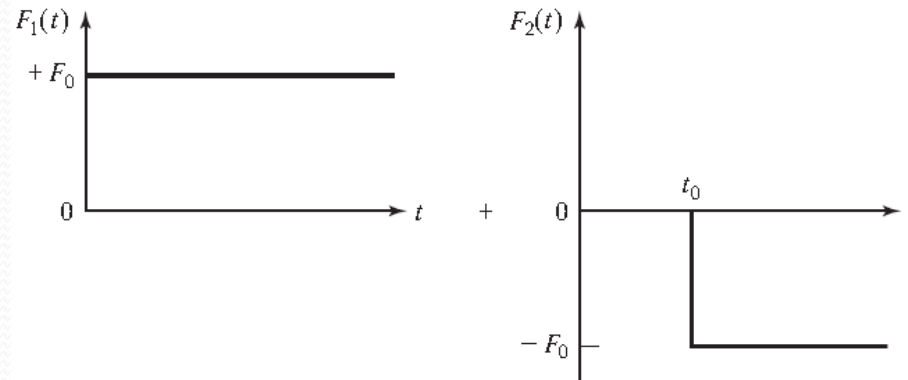
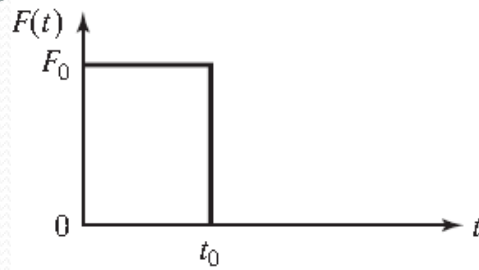
$$x(t) = \frac{F_0}{k} [1 - \cos \omega_n t] \quad (\text{E.3})$$

Equation (E.3) is shown graphically in Fig. 4.10(d). It can be seen that if the load is instantaneously applied to an undamped system, a maximum displacement of twice the static displacement will be attained—that is, $x_{\max} = 2F_0/k$.



EXAMPLE 4.11

If the compacting machine shown in Fig. 4.10(a) is subjected to a constant force only during the time $0 \leq t \leq t_0$ (Fig. 4.12a), determine the response of the machine.



Solution: The given forcing function, $F(t)$, can be considered as the sum of a step function $F_1(t)$ of magnitude $+F_0$ beginning at $t = 0$ and a second step function $F_2(t)$ of magnitude $-F_0$ starting at time $t = t_0$, as shown in Fig. 4.12(b).

Thus the response of the system can be obtained by subtracting Eq. (E.1) of Example 4.10 from Eq. (E.1) of Example 4.9. This gives

$$x(t) = \frac{F_0 e^{-\zeta \omega_n t}}{k \sqrt{1 - \zeta^2}} \left[-\cos(\omega_d t - \phi) + e^{\zeta \omega_n t_0} \cos\{\omega_d(t - t_0) - \phi\} \right] \quad \phi = \tan^{-1} \left(\frac{\zeta}{\sqrt{1 - \zeta^2}} \right)$$

Blast Load on a Building Frame

EXAMPLE 4.13

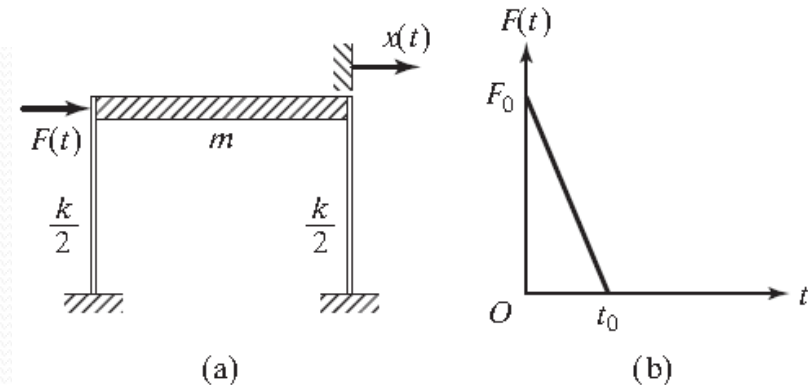
A building frame is modeled as an undamped single-degree-of-freedom system (Fig. 4.14(a)). Find the response of the frame if it is subjected to a blast loading represented by the triangular pulse shown in Fig. 4.14(b).

$$F(\tau) = F_0 \left(1 - \frac{\tau}{t_0} \right) \quad \text{for } 0 \leq \tau \leq t_0$$

$$F(\tau) = 0 \quad \text{for } \tau > t_0$$

$$x(t) = \frac{1}{m\omega_n} \int_0^t F(\tau) \sin \omega_n(t - \tau) d\tau$$

$$x(t) = \frac{F_0}{k} \left[1 - \frac{t}{t_0} - \cos \omega_n t + \frac{1}{\omega_n t_0} \sin \omega_n t \right]$$



Response during $t > t_0$: Here also we use Eq. (E.1) for $F(\tau)$, but the upper limit of integration in Eq. (E.3) will be t_0 , since $F(\tau) = 0$ for $\tau > t_0$. Thus the response can be found from Eq. (E.7) by setting $t = t_0$ within the square brackets. This results in

$$x(t) = \frac{F_0}{k\omega_n t_0} \left[(1 - \cos \omega_n t_0) \sin \omega_n t - (\omega_n t_0 - \sin \omega_n t_0) \cos \omega_n t \right] \quad (\text{E.9})$$